

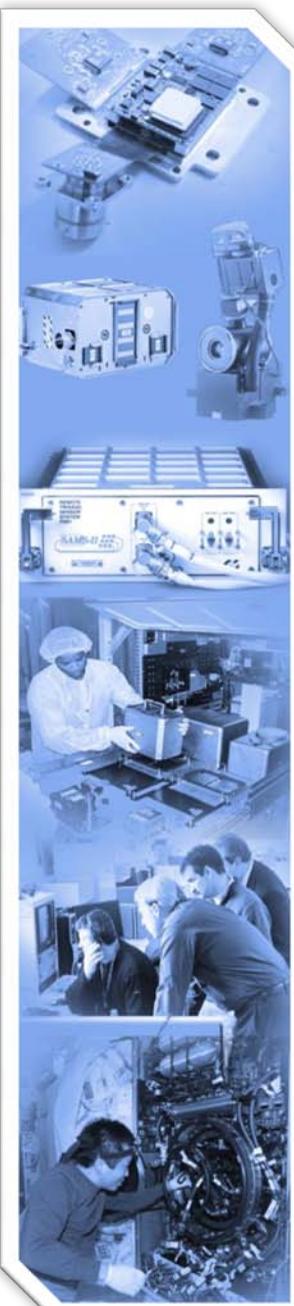
Supply Chain 2011 Moving Innovative Product into NASA Mission Projects Carlos Grodsinsky, Ph.D. V.P. Technology

Technology
without Boundaries



YOUR STRATEGIC DIVERSIFIED SMALL BUSINESS (MO-SDB) PARTNER

Management | Design | Analysis | Integration | Manufacturing | Test | Operations



Presentation Outline

- Introduction – Understanding the NASA Market Place
- NASA Mission Centers and Mission Needs
- Technology Driven Competitive Product/Services within the NASA Market Segment
- An Innovative Product NASA Success Story - Magnetospheric MultiScale (MMS) Mission Acceleration Measurement System (MMSAMS)



MO-SDB

AS9100B certified

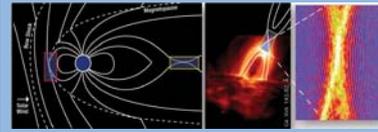
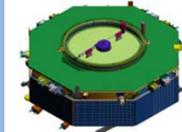
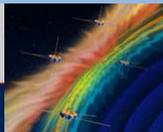
**DoD Facility
Clearance**

**DCAA Approved
Forward Pricing**

Experienced Team

**Award Winning
Capabilities**

**Headquartered in
Cleveland Ohio**





Understanding the NASA Market Place

Know Your Customer and the Products and Solutions they Need to Successfully Complete Their Mission, Objectives, Goals

- **NASA Supply Chain is Segmented into Research and Mission Centers**
 - **Mission Center Needs**
 - Flight Systems, Subsystems, Engineering Services (System Engineering and Integration, Hard Engineering, Engineering Processes)
 - **Research Center Needs**
 - Each Research Center has their core expertise which dictates the types of Engineering Services /Products and Development tools and Systems
 - Research Centers Support Mission Centers and also are responsible for demonstration programs/projects

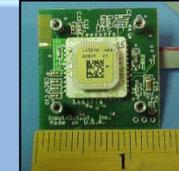
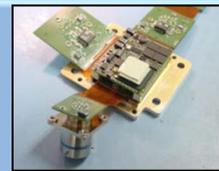
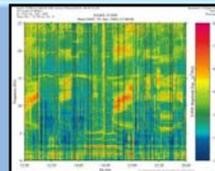


Product/Service Solutions

Technical : Product

Past Performance: Experience, Core Competence, History

Cost: Competitive Pricing



Understanding the NASA Market Place (Cont.)

Identifying Opportunities in Relation to NASA Needs

- All procurement and solicitation opportunities are published through the NASA procurement sites
 - <http://prod.nais.nasa.gov/cgi-bin/nais/index.cgi>
- All products or services follow three main criteria for competitive award:
 1. Technical Solution – Product or System Capability to Requirements
 2. Past Performance – Experience, Capability and Evidence of Relevant Past History
 3. Competitive Cost Solution (Competitive Range and Competitive to Scope of Mission and Program/Project Budget)



Product/Service
Solutions

Technical : Product

Past Performance:
Experience, Core
Competence,
History

Cost: Competitive
Pricing





Understanding the NASA Market Place (Cont.)

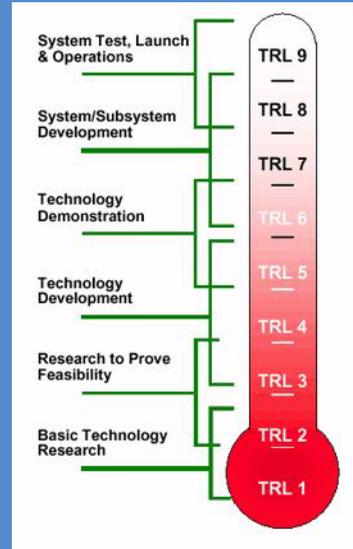
Know your Customers Organization and Objectives

- NASA Unmanned Missions are driven by independent decadal science reviews, competed missions, or specific missions against four broad scientific pursuits
 - (Earth Science, Planetary Science, Heliophysics, and Astrophysics)

http://science.nasa.gov/media/medialibrary/2010/03/31/Science_Plan_07.pdf

Examples -

- Planetary Science Decadal Survey
- Earth Science Decadal Survey – the source of Announcement of Opportunities which are published to science community – MMS came from one of these AO's
- NASA Research Centers support these missions by working on technologies that enable these missions
 - Technology Roadmaps are developed to provide requirements and needs for these mission enabling technologies – standard formulation of Technology Readiness Levels (TRL 1 through 9)



Product/Service Solutions

Technical : Product

Past Performance: Experience, Core Competence, History

Cost: Competitive Pricing





Technology Driven Product/Services

Unique Services Aligned to NASA Missions and Technology Needs

- Engineering Services – Analysis, System Engineering and Integration, Configuration Management, Hardware and Software design, development , fabrication and integration
- Safety and Payload Integration
- Ground Processing and Ground Systems
- Operations

Innovative Products driven by Technology Roadmap Gaps

- Utilize SBIR and other sources of development funds
- Directed Research and Development under NASA Infrastructure or Prime Delivery Order contracts

Innovative Technology Driven Product Example

- Inertial Accelerometer Flight Hardware - MMSAMS – Heritage from Space Acceleration Measurement System and Microgravity Acceleration Measurement System (SAMS/MAMS)



RESEARCH AND TECHNOLOGY Driven

Products and Services

Concept Definition

Detailed Design

Engineering

Manufacturing

Integration

Testing

Verification

Certification

Maintenance

Ground Processing

Training

Operations Services





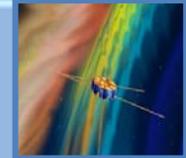
A Technology Driven Innovative Product Success Story MMSAMS

MMS Required Unique Set of Inertial Measurement Requirements to Provide Formation Flying information for MMS Satellites

- **Technical - Technology Driven Solution**
 - High Resolution High Accuracy Performance
 - High Reliability Design and System
 - Diverse set of Communication Requirements
- **Past Performance – Heritage in Space Flight High Reliability**
 - Past Performance providing Space Flight Hardware
 - Space Flight Heritage in Design
 - Experience (core expertise in acceleration measurement and flight dynamics)
- **Competitive Cost – Competitive Award and within Mission Budget for sub system**



Technology Knowhow and Mission Requirements





Space Acceleration Measurement System (SAMS)

Objective:

Provide acceleration measurement systems that meet the requirements of the researchers on board the ISS and provide vehicle health data. SAMS measures the acceleration environment in the 0.01 to 400 Hz range for payloads.

Longest running Near Earth Orbit Inertial Measurement (Acceleration package) over 20 STS, MIR and ISS missions

- A SAMS unit is currently displayed in Smithsonian Air and Space Museum as example of NASA Microgravity Science Payload with over 20 missions inclusive of MIR operation

Current Space Acceleration Measurement and Microgravity Acceleration Measurement Systems (SAMS) on-board ISS for 8 years in April 2010 (continuous operation)

- MAMS – inertial sensor resolution 2-3 nano-gs
- SAMS – inertial sensor package resolution .1 micro-gs
- Corrected Bias error for SAMS, less than .5 micro-g
- MEMS based inertial measurement unit designs for sub .01 degree per second angular resolution, and sub-100 nanog/root Hertz linear acceleration resolution



Space Acceleration Measurement System

SAMS keeps going and going . . .

1991	SL-1	SL-1	
	STS-43	STS-100 to ISS	2006 ISS
1992	IML-1	2002 ISS	2007 ISS
	USMP-1	2003 ISS	2008 ISS
1993	USMP-4	STS-107	2009 ISS
	SPACELAB-1	2004 ISS	2010 ISS
	SPACELAB-2	2005 ISS	
	USMP-2		
	IML-2		
	Mir Space Station		
	STS-66		
1995	Mir Space Station		
	USMP-3		
	IML-2		
	Mir Space Station		
	USMP-3		
	LVS		
	STS-79/Mir-4		
1997	Mir Space Station		
	Mir-1		
	Mir-2		
	USMP-4		
1998	Mir Space Station		
	STS-89/Mir-8		
	STS-91 (Mir Retrieval)		
	STS-95 (HST)		

22x2010
Microgravity Environment Project
ISS Research Program
NASA Glenn Research Center, Cleveland, Ohio

20 years!

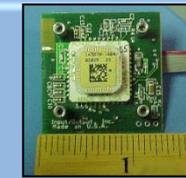
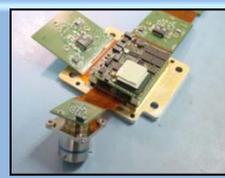
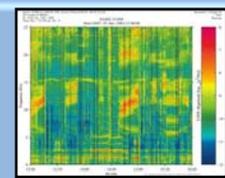
SAMS Has Characterized Every Current NASA Manned Spacecraft in the Past 20 Years

ISS, MIR, SHUTTLE, SPACELAB, SPACEHAB

SAMS Unit On Display at the Smithsonian Air & Space Museum



MIR SAMS Unit on Display At the Smithsonian Institute Steven F. Udvar-Hazy Center





Microgravity Acceleration Measurement System (MAMS)

MAMS

MAMS Currently Supports ISS Reboost Approximately Every 2 to 3 Months

Supports ISS Loads and Dynamics Group



Objective:

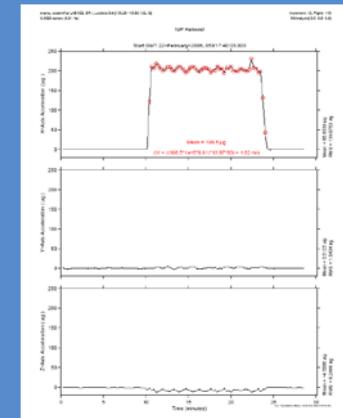
One of the major goals of the International Space Station (ISS) is to provide a quiescent low-gravity environment to perform fundamental scientific research. However, small disturbances aboard the ISS impact the overall environment in which experiments are being performed. Such small disturbances need to be measured in order to assess their potential impact on the experiments. The Microgravity Acceleration Measurement System (MAMS) is used onboard the ISS to do just that.

Provides acceleration measurement system that measures the Quasi steady and vibratory acceleration data in the 0.00001 to 100 Hz frequency range on board the International Space Station (ISS) vehicle.

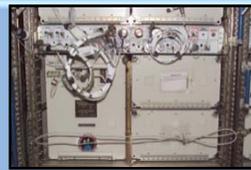
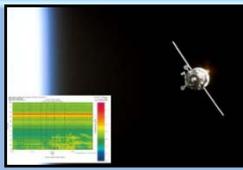
- MAMS will measure the acceleration environment for ISS structures as requested and provide data to vehicle (ISS and/or Shuttle) for health and dynamic analyses.
- MAMS was developed to operate with minimum crew interaction, and can be commanded with ground commands.
- MAMS supports the ISS reboosts, dockings, and exercise.
- MAMS current on board mass is 53.1 kg, with a volume of 0.154 cubic meters.
- Future applications could include new spacecraft docking assessments and rocket stage separation characterization



ISS On-orbit Front Panel



MAMS ISS Reboost Data

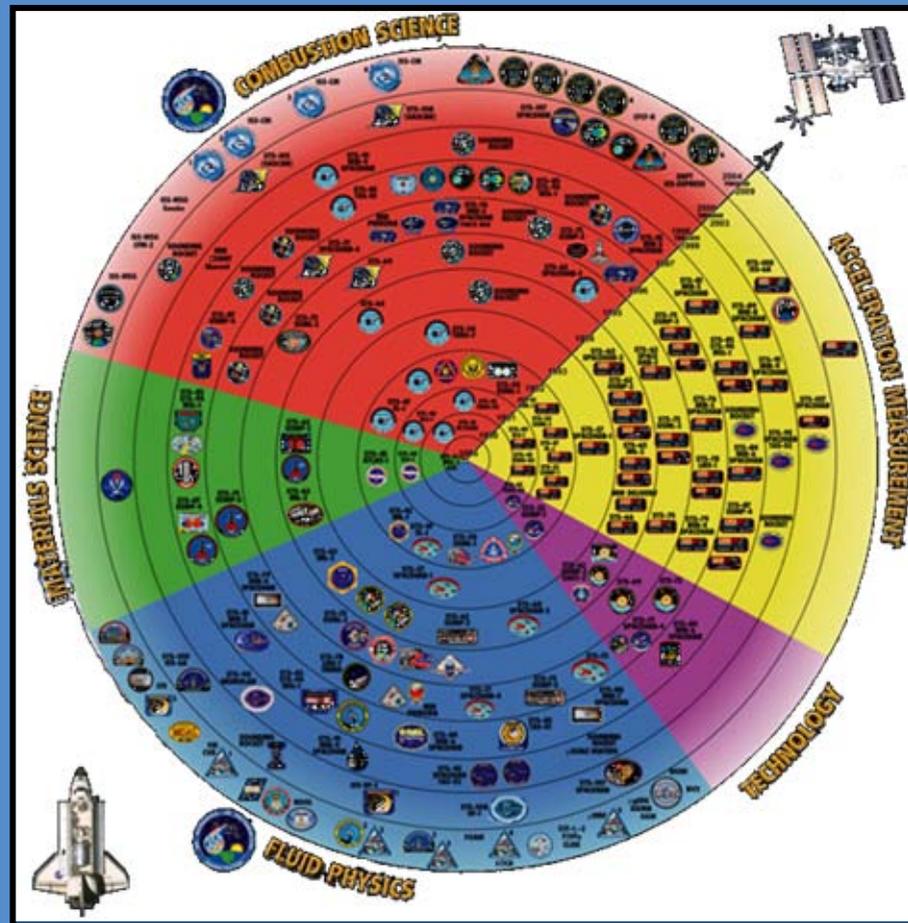




Spaceflight Hardware/Software Heritage

For over 20 years, the ZIN engineering team has partnered with NASA management, scientific experts and industry to manage and develop space flight technology development and scientific systems, from concept definition, design, development, and fabrication to system assembly, integration, test, launch, operations and return.

- ➔ ZIN has extensive experience developing NASA human-rated space flight systems as well as experience with Sounding Rockets and Drop-Towers. Our engineering Team has designed, fabricated and operated over 133 human-rated space flight payloads with thousands of hours of space flight logged on the shuttle, MIR, and ISS.



Space Flight Hardware Experience

Concept Definition

Detailed Design

Engineering

Manufacturing

Integration

Testing

Verification

Certification

Maintenance

Ground Processing

Training

Operations Services



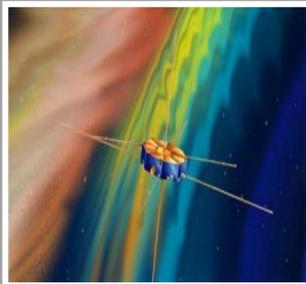
Three-Axis
Accelerometer System

SAMS Heritage

Will Measure Changes
in Spacecraft Velocity

Four MMS
observatories

Will Fly Tetrahedral
formation



Magnetospheric Multiscale Mission (MMS)

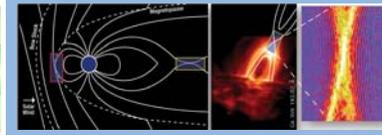
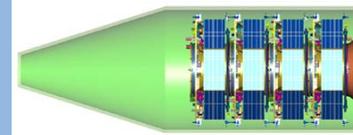
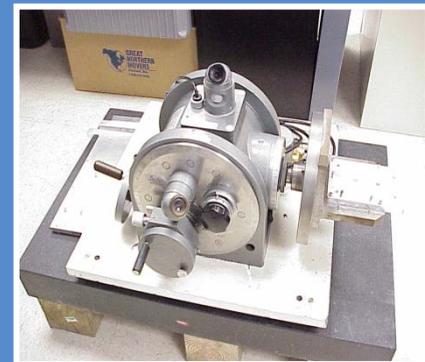
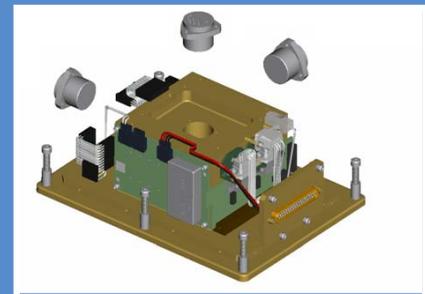
The Magnetospheric Multiscale (MMS) mission is the fourth mission of the Solar Terrestrial Probe (STP) program of the National Aeronautics and Space Administration (NASA)

The MMS mission will study magnetic reconnection in the Earth's magnetosphere. The four MMS observatories will be required to fly in a tetrahedral formation in order to unambiguously determine the orientation of the magnetic reconnection layer.

ZIN is working on providing the Three-Axis Accelerometer Systems based on our NASA developed Space Acceleration Measurement System (SAMS).

The Accelerometer System will be used to accurately determine changes in spacecraft velocity to precisely maintain formation of the four spacecraft and to provide velocity feedback during apogee raising maneuvers.

Demonstrated capability to integrate the corrected acceleration, at 1000 Hz, by applying Delta V calculations.

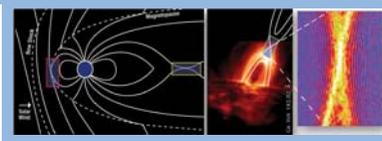
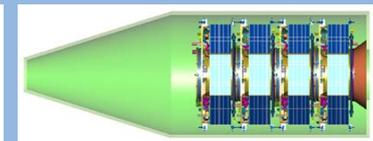
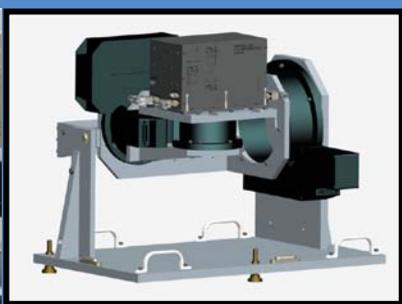
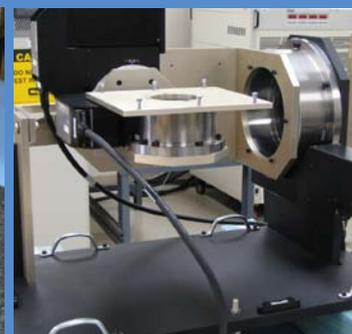


Magnetospheric Multiscale Mission (MMS)

Three-Axis Accelerometer System
 SAMS Heritage
 Will Measure Changes in Spacecraft Velocity
 Four MMS observatories
 Will Fly Tetrahedral formation



Item #	Description	Reference	Category	Quantity	Delivery Date
1	Accelerometer System Engineering Test Unit	SOW 4.2	A	1	6/30/2011
2	Accelerometer System Flight Unit	SOW 4.2	A	4 units	Flight 1: 2/3/2012 Flight 2: 5/11/2012 Flight 3: 8/12/2012 Flight 4: 11/23/2012
3	Accelerometer System Connector Savers	SOW 4.3	A	5 sets	With items #1 and #2
4	Accelerometer System Supporting Hardware	SOW 4.4	A	5 sets	Closeout Caps Transport Caps and ESD Caps: With Items #1 and #2 Mating Connectors: 6 months after Award of Contract Drill Template: 18 months after Award of Contract





Company Overview

ZIN Technologies Inc. (ZIN) a minority owned small disadvantage business (MO-SDB) has provided advanced aerospace / biomedical engineering services and products for NASA, DoD, centers for scientific research and aerospace corporations.

With over five decades of experience, our capabilities are continually being improved to stay on the leading edge of rapidly evolving technologies that meet the needs of our customers.

ZIN specializes in the seamless and transparent progression between concept, detailed design, engineering, manufacturing, integration, testing, verification, certification, maintenance and ground, air and space operations.

- 175+ person organization consists of scientists, engineers, designers, and technicians experienced in managing complex programs and technical requirements.**
- ZIN has extensive experience developing NASA human-rated space flight systems. Our engineering Team has designed, fabricated and operated over 133 man-rated space flight payloads with thousands of hours of space flight logged on the shuttle, MIR, and ISS.**
- 50,000 square feet AS9100 certified facility**
- Robust/Certified Government support systems and DCAA approved forward pricing**
- DISC Central Verification Activity can provide information regarding our current DOD facility clearance status to the appropriately requesting parties at (888)282-7682**
- We have received numerous quality and performance awards including recognition as a 2003 and 2004 NASA George M. Low Quality Award Finalist**



MO-SDB

AS9100B certified

**DoD Facility
Clearance**

**DCAA Approved
Forward Pricing**

Experienced Team

**Award Winning
Capabilities**

**Headquartered in
Cleveland Ohio**



Engineering &
Technical Services

Hardware & Software
Development

Science/Exploration
Technologies

Space Technology
Demonstration

Space Systems
Integration & Operations

Human Health &
Medical Devices



AS9100B Certified Facilities

ZIN Technologies occupies 50,000 square feet in its AS9100 certified corporate headquarters in Cleveland, Ohio.

Our facility includes a separate flight hardware assembly area with an electronic buildup capability, all located adjacent to our corporate headquarters. The assembly area provides 20,000 square feet of work space and is equipped with a circuit board fabrication laboratory where circuit boards can be processed, cleaned and conformal-coated.

Our engineers and technicians routinely build, test and fabricate ground and flight hardware including prototype analog and digital instrumentation using our oscilloscopes, microscopic inspection stations, soldering stations, precision voltage sources, digital multimeters, logic analyzers, & dynamic signal analyzers.

Hardware Assembly Area - Clean room & laminar flow workspace are available for contamination-sensitive components. Our laboratories can be operated as a class 100,000 visibly clean room with a humidity control system for ESD prevention. Laser-safe work areas and a thermal test chamber are available.

Machine Shop - ZIN has an in-house capability and 5000 sq ft work area for fabrication of prototype hardware and limited quantities of deliverable hardware.

Property Management - ZIN has an approved procurement department, Government Property Control System certified by the Defense Contract Management Agency, shipping and receiving area, and a climate controlled bonded storage area.

Manufacturing Work Order Process - Our manufacturing systems include material and process control plans, EEE parts control, contamination allowance and control, risk management, PRACA system and engineering review boards.



Admin & Engineering



Laboratories



Manufacturing



Assembly



Integration & Test





Experienced Engineering and Management

Experienced Technical Resources:

- Self-directed motivated work team capable of hardware and end item deliverable accountability
- Project-focused development teams, defined milestones & deliverables
- Long term experience with hardware and software development, launch, mission and flight centers

Established Project Mgmt Infrastructure:

- Mature Business and Project Management infrastructure capable of supporting Large Business Management Systems and reporting requirements
- Earned-Value Management systems
- Configuration Management Systems
- Product Assurance Systems
- Verification and Requirement Management Systems
- Scheduling, monthly reporting and administration systems

Experienced Management Team:

- Experienced with Small "turn-key" contracts with small budgets & tight schedules
- Decades Of Experience Working with Customers Such as NASA, DoD, and Fortune 500 Companies
- We are an experienced NASA Prime Contractor with demonstrated responsibilities for large manned space flight programs
 - An example is the \$110 Million International Space Station Fluids and Combustion Facility (multi-year and inclusive of sustaining engineering and long-term on-orbit operations)
- Experienced management of multiple subcontractors including large prime contractors, SDBs, 8As, Hub Zone and woman owned certifications

Established Engineering Support Systems:

- AS 9100 Quality System
- Processes controlled via Procedures and Work Instructions
- Manufacturing Work Order System
- Configuration Management expertise for NASA and DOD programs
- Hardware traceability including waivers, deviation, engineering changes through hardware certification
- Experience with ZIN, Konfig and Windchill CM products
- Change Control Board Processes
- Engineering Review Board Processes
- Risk Management Database and PRACA Systems
- Certified-QA inspectors for electronic and mechanical workmanship
- Parts & material tracking including material certifications, material test results, certificates of conformance, lot numbers, and other special documentation

ESTABLISHED
ENGINEERING AND
MANAGEMENT
SYSTEMS

Experienced Technical
Resources and
Managers

Large and Small
Project experience

Collaboration and
Teamwork

Established Support
Systems

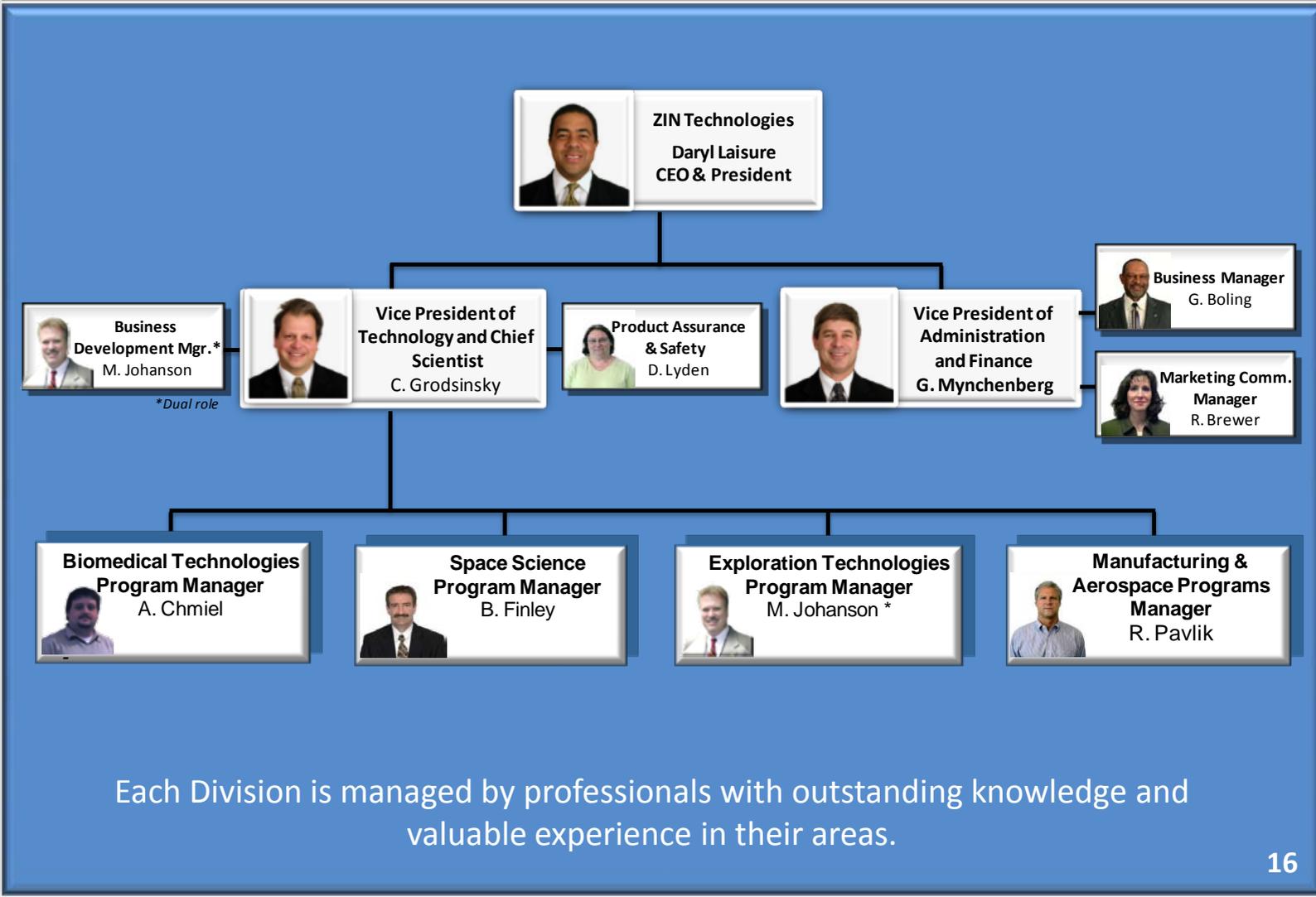
Turn Key or Task Order

Business Management
Systems and Reporting





Organized to Meet Customer Requirements



Each Division is managed by professionals with outstanding knowledge and valuable experience in their areas.

EXPERIENCED
MANAGEMENT TEAM

Experienced Technical
Resources and
Managers

Business Management
Systems and Reporting

Project Focused Teams

Collaboration,
Partnership and
Teamwork

Large and Small
Project Experience

Turn Key or Task Order

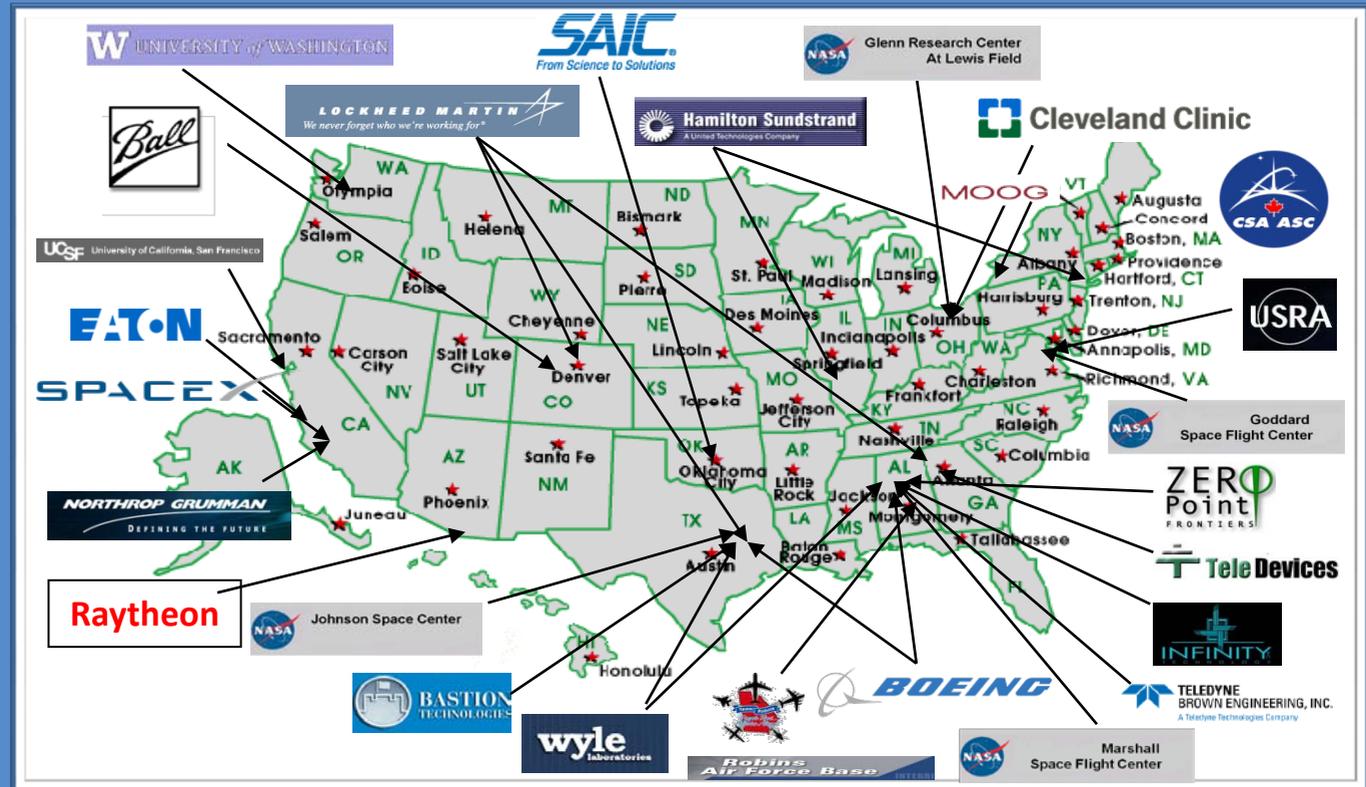
Self-directed
motivated Team
capable of
hardware and end
item deliverable
accountability





Customers/Partnerships

Based on our experience as a NASA prime contractor, we are capable of providing complex aerospace products and engineering support with confidence. By working closely with you over time to integrate into your business and understand your full range of requirements, we hope to be your technology solution for the long term.





Control Systems

Processing

Data Acquisition

Data Storage

Image Processing

Digital & Analog

Programmable

Health & Status

COTS & Custom

Communications



ZIN Developed Avionics
Package on ISS

Examples Of ZIN Developed Custom Avionics

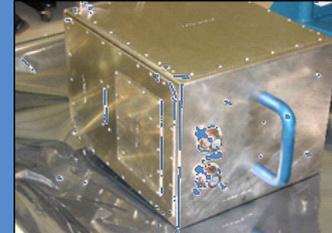
Data Acquisition • Health & Status • Control • Processing & Storage



Fluids Science Avionics Package

Serves as the control and data acquisition system for a science payload: RS -422, 2 channels, A/D, D/A, DIO, Motion control, Analog video, CAN bus control of diagnostics and PI Hardware

Two 36 GB hard drives
P3-500mhz -RAM
Compact PCI: 8 Slots, 32 bit
28V @ 77 Watts



FOMA Control Unit

Serves as the control and data acquisition system for the FOMA:

Provide power conditioning
Controls all gas blending
Controls gas chromatograph for chamber sampling.
Provides hardware status to the IOP via Ethernet and CAN bus
6u VME - 7 Slots
28V @ 92 Watts

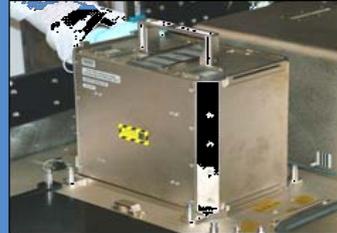


Image Processing and Storage Unit

Stores digital image data from a camera
Image analysis, processing and reduction
Control signals to camera and Illumination packages
IEEE 1394, FireWire Interface

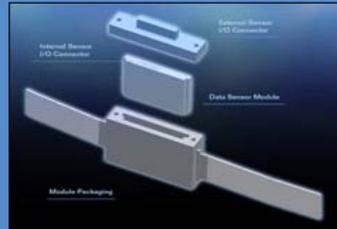
Analog video, RS -170A, input/output
Two 36 GB hard drives
P3-500mhz -RAM
6 slot Compact PCI, 64 bit custom designed backplane
28V @ 100 Watts



Space Accelerations Measurement System

ISS Microgravity Environment System and Subsystem Test Verification and Analysis

COTS PC/104 Embedded Controller
QNX Real -Time OS Multi -Tasking, Preemptive Scheduling, and Fast Content Switching
Custom Triaxial Sensor Head, Closed -Loop Fiber Optical Roll -rate Gyro, and Custom Very Low Frequency Sensor



Bio Watch

Dynamic Data Acquisition
16 channels of data acquisition, w/ 24 bit resolution
10-4000 Hz Sample Rate (32,000 Hz max throughput)
120 dB dynamic resolution
Programmable individual channel filtering
Data Communication - Investigation into using commercial wireless technologies in space vehicles
Achieve bandwidth to support aggregate 32 kHz



Input/Output Processor

Main Fluids & Combustion Facility controller and data acquisition system.

Video - 16x8 video switch,
Ethernet Switch, Sync Bus
6u VME & Compact PCI
4 Slot 6u VME, 1 Slot compact PCI
28V @ 120 Watts



CVB Control Box

Serves as avionics control interface for Fluids Science Avionics Package through a data/power cable

28V and a maximum current of 4 Amps.
Compatible with FSAP Compact PCI Interfaces



PCS Avionics Section

Ethernet (10Base -T) Communications
PCI/ISA Form Factor
Analog -to-Digital Conversion w/Conditioning, 8 -axis Motion Control
Digital I/O, 16 -Bit Digital Framegrabber Interface
RS-170 Interface
Removable SCSI Hard Drives
Laser Controller Interface (2)





Examples Of ZIN Developed Hardened Diagnostics

Illumination • Imaging • Diagnostics/Sensors • Communications



White Light Package

Provides uniform, broad band lighting
Two independent light engines
Adjustable intensity
Fiber Optic Quick disconnects
Mounted to rear of bench, quick connect/disconnect of fiber bundles
C-515C Microcontroller
151 Watts



Nd:YAG Laser

Provides a laser source for various diagnostic techniques such as Particle Image Velocimetry
532 nm, 150mw Output power
Analog control of laser functions
Laser output power monitoring
C-515C Microcontroller
26 Watts



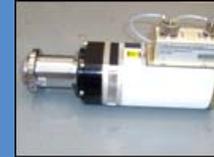
Color Camera Package

24 Bit, 3 chip CCD
1/3 inch array, 768 X 494 pixel
RS 170C output (30 FPS)
Remote and interchangeable head allowing for in-situ calibration with controller
C-515C Microcontroller
35 Watts



Surveillance Camera

Low resolution observation
Point Grey Firefly™ Monochrome Camera
IEEE-1394 digital
1/3 CCD Sony, VGA 640x480 format - 3.75, 7.5, 15 and 30 FPS
Focal lengths of 4mm, 6mm, or 8mm



QImaging Camera

High resolution imaging QImaging Retiga 1300C
1280 x 1024 pixels, 12 bit monochrome
6.7 μm x 6.7 μm pixel size
Dark Current 0.15e⁻/pix/s cooled Selectable Region of Interest (ROI)



HiBMs

High Bit-Depth Multi-spectral Diagnostic Package
Field of View: 50 mm square or 90 mm diameter
Spectral filtering provided with the use of a Liquid Crystal Tunable Filter.
12 bit per pixel output
4095 gray levels
frame rates of 7.5, 15 or 30 fps
PC 104
54 Watts



Low Light Level UV

field of view : 50mm or 80mm
4095 gray levels
12 bits per pixel output
frame rate 7.5, 15, 30 fps in 1x1 mode (1kx1k resolution)
7.5, 15, 30 and 54 fps in binning mode (2x2, higher sensitivity less resolution)
Low Light Level in the Ultra Violet spectrum.
PC 104
49 Watts

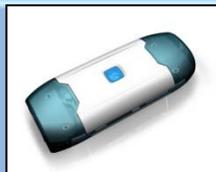


Illumination Control Module

Illumination source used for general backlighting and Soot Volume Fraction experiments
5 mW coupled power minimum from diode Illumination Source
Spectral Bandwidth: 10 nm maximum at 50% points
90 mm beam diameter
Wavelength 675 ± 15 nm
PC 104
24 Watts



ZIN Developed Light Microscopy Module (LMM) an Automated Microscope Diagnostic Tool on ISS





Experienced in Exploration Technology Demonstration

- Project & Program Management
- System Engineering
- Electro-Mechanical & Electrical Design
- Embedded System Design & Development
- Fluids Systems
- Structural & Thermal Analysis
- Digital & Analog System Design
- Environmental Testing
- Product Assurance
- Safety & Material Engineering
- Power Conversion Systems
- Optical Diagnostics
- Embedded Software Development
- Remote Biomedical Monitoring
- Bioastronautics Flight Hardware
- Flight Hardware Integration and Operations



Habitation



Pre-Positioned Propellants



Power



ISS Research



Resource Identification



Biomedical Countermeasure

Technology Options

- ☺ Power Systems
- ☺ Propulsion Systems
- ☺ Life Support Systems
- ☺ Air Revitalization
- ☺ Water Reclamation
- ☺ Thermal Management
- ☺ Sensors & detectors
- ☺ Radiation Monitoring
- ☺ Radiation Hardened Electronics
- ☺ Nano Thin Film Processing
- ☺ Low Gravity Environments
- ☺ Composites Structures

Supporting Research

- ☺ 133 Man-Rated Science payloads
- ☺ Fluids & Combustion Facility
 - Fire Safety
 - Advanced Life Support
 - Water Reclamation
 - BioScience
 - Contaminants Identification
 - Microorganism identification
 - Human Health
- ☺ Human Health, Bioastronautics/ Infomatics
 - EMG Data Analysis Tools
 - Astronaut Exercise
- ☺ Requirements & Systems Engineering



Crew Transport



Launch



Crew Support



Landing Systems



Comm/Nav



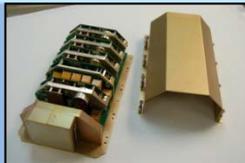
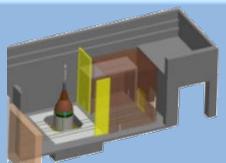
Surface Mobility

Transit & Launch Systems

- ☺ Power Systems
- ☺ Propulsion Systems
- ☺ Thermal Protection
- ☺ Cryogenics Management
- ☺ Fluids handling Systems
- ☺ Structures
- ☺ Life Support Systems
- ☺ Radiation Hardened Electronics
- ☺ Sensors & Detectors
- ☺ Communications Systems
- ☺ Spaceflight Hdw & Sw
- ☺ Integration n& Operations Services
- ☺ Space Structure Interactions
- ☺ Low Gravity Environments

Orbital & Surface Systems

- ☺ Power Systems
- ☺ Propulsion Systems
- ☺ Thermal Protection
- ☺ Fluids Handling Systems
- ☺ Life Support Systems
- ☺ Radiation Hardened Electronics
- ☺ Sensors & Detectors
- ☺ Communications Systems
- ☺ Low Gravity Environments
- ☺ Spaceflight Hdw & Sw
- ☺ Integration & Operations Services
- ☺ Automated Operations

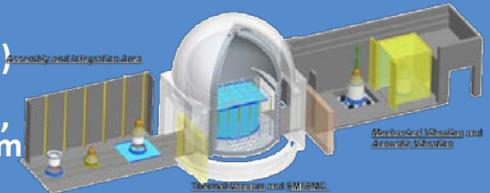




Crew Vehicle Spacecraft Technology Demonstrations

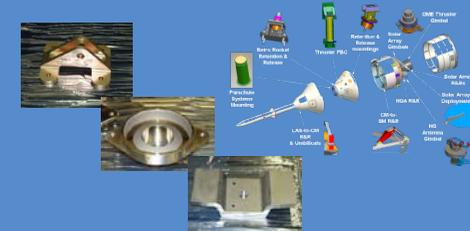
ORION Vibro-Acoustic Test Capability at NASA Plum Brook Station Space Power Facility

- This new capability includes a Mechanical Vibration Facility (MVF) and a Reverberant Acoustic Test Facility (RATF). In support of the GRC-based contract, ZIN is responsible for the design, installation, test and commissioning of the High Speed Data Acquisition System (HSDAS).



ORION Mechanisms and Structural Detailed Parts

- Under this contract with the Orion Prime Contractor Lockheed Martin, ZIN with our partner Infinity Technologies manufactured, inspected and delivered mechanisms and simulator detailed parts to be used in early ground testing and any initial flights of the Orion Crew and Service Module.



ORION, Active Cooling Technology Development Testing

- In support of prime contractor Hamilton Sundstrand, ZIN performed testing to simulate the active cooling of the Orion Crew Vehicle electronics during the reentry phase of Orion spacecraft.



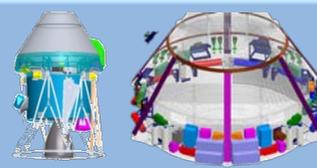
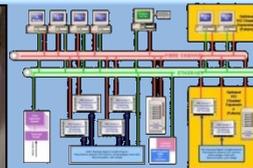
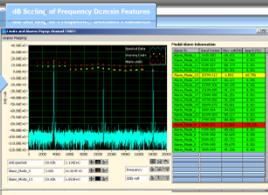
ORION, Portable Fire Extinguisher

- In support of prime contractor Hamilton Sundstrand, ZIN performed requirements definition and development program. ZIN assisted in foam selection, nozzle design and test of engineering model hardware. Design allows for water mist as an alternative.



CEV (ORION), Phase I Team

- In support of prime contractor Northrop Grumman/Boeing, ZIN contributed to the Electrical Power System design. We utilized our expertise in power systems to create a detailed model for spacecraft energy balance. This required a comprehensive database of electrical loads with provisions to model on/off/duty cycle status that varies with time.





EXPERTISE INCLUDES
MULTIPLE LAUNCH
VEHICLES

Sounding Rockets

Space Shuttle

Soyuz

Progress

ATV

HTV

Commercial

INTEGRATE AND
OPERATE IN MANY
ON-ORBIT CARRIERS

ISS

Glovebox (MSG)

Fluids & Combustion
Facility (FCF)

EXPRESS Rack

Exterior Pallet

Commercial

Space System Integration & Operations Expertise

Payload Analytical Integration

- Requirements Definition
- Manifesting
- ICD Development
- Payload Verification

Mission Analytical Integration

- Safety Certification
- Mission Requirements Definition
- Mission Design
- Mission Design Verification

Mission Training

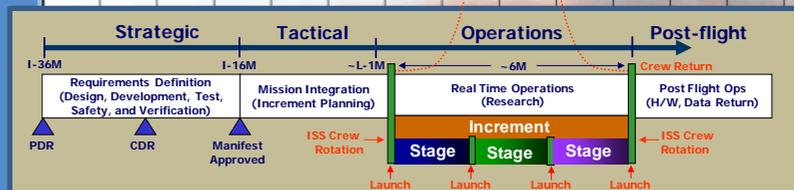
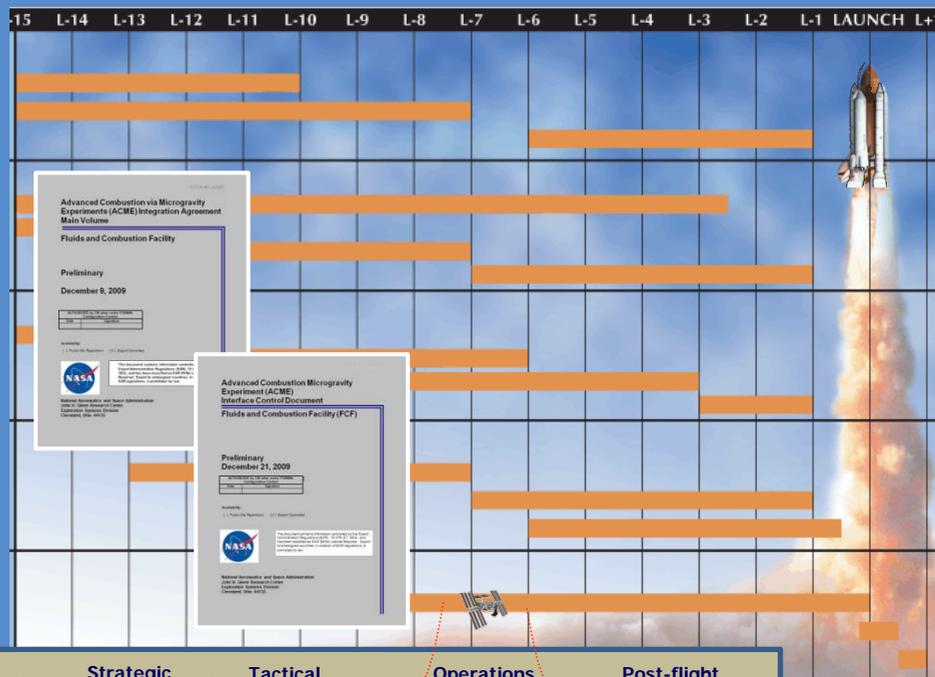
- Ground and Flight Personnel
- Mission Training Plan
- Familiarization Training
- Hands-on Training
- Joint Integrated Simulations

Hardware Physical Integration

- Requirements Development
- Carrier Integration and Test
- Payload Integration and Test

Mission Operations

- Procedure Development
- Real-Time Flight Operations
- Daily Planning
- Mission/Post-Flight Analysis



The ZIN Team has ISS Payload Integration facilitates support expertise from concept through on-orbit operations





ZIN Technologies Summary

Current Contract Work Scope

- ISS FCF sustaining engineering and operations;
- Unique ISS flight investigations in combustion science, fluid physics, materials science and accelerometry;
- Ground-based and flight research investigations in exercise countermeasures equipment, medical devices and biosensors for long duration space missions;
- Advanced EVA (extravehicular activity) technologies in power, communications, avionics and infomatics in support of lunar surface operations;
- Instrument packages, and power, propulsion and sensor subsystems for Earth science, lunar exploration and other space science missions aboard spacecraft and airborne science platforms;
- Advanced technology development and demonstrations in the areas of power, propulsion, space communications systems and subsystems, lunar surface and in-situ resource applications, spacecraft fire safety, human research and exploration medical capability;
- STS/ISS and other space flight systems and subsystems;
- Advanced Exploration Technology Development - Ares, Orion and Altair systems and subsystems;
- Satellite Subsystem development;
- Commercial Spacecraft.

Experienced Minority Owned Small Disadvantaged Business (MO-SDB) Partner

- ZIN is an experienced developer of ground and flight systems and components for manned and unmanned aerospace applications. Our engineering expertise includes Electro-Mechanical Systems, Instrumentation, Power, Data Acquisition, Software Development and Power Conversion products.
- ZIN specializes in the seamless and transparent progression between concept, detailed design, engineering, manufacturing, integration, testing, verification, certification, maintenance and operations.

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